Manual Design Procedure for Columns and Walls with Biaxial Bending (ACI 318-11/14/19)





PM at 23.0 [deg]
MM at $\mathrm{P}=1846.0$ [kip]

## Manual Design Procedure for Columns and Walls with Biaxial Bending (ACI 318-11/14/19)

Biaxial bending of columns occurs when the loading causes bending simultaneously about both principal axes. The commonly encountered case of such loading occurs in corner columns. Corner and other columns exposed to known moments about each axis simultaneously should be designed for biaxial bending and axial load.

A uniaxial interaction diagram defines the load-moment strength along a single plane of a section under an axial load $P$ and a uniaxial moment $M$. The biaxial bending resistance of an axially loaded column can be represented schematically as a surface formed by a series of uniaxial interaction curves drawn radially from the P axis. Data for these intermediate curves are obtained by varying the angle of the neutral axis (for assumed strain configurations) with respect to the major axes.

The difficulty associated with the determination of the strength of reinforced columns subjected to combined axial load and biaxial bending is primarily an arithmetic one. The bending resistance of an axially loaded column about a particular skewed axis is determined through iterations involving simple but lengthy calculations. These extensive calculations are compounded when optimization of the reinforcement or cross-section is sought.

This example demonstrates the determination of a column section capacity ( $\mathrm{P}_{\mathrm{n}}, \mathrm{M}_{\mathrm{nx}}$ and $\mathrm{M}_{n y}$ ) required to resist the following factored load and moments: $\mathrm{P}_{\mathrm{u}}=1200 \mathrm{kips}, \mathrm{M}_{\mathrm{ux}}=300 \mathrm{kip}-\mathrm{ft}$, and $\mathrm{M}_{\mathrm{uy}}=125 \mathrm{kip}-\mathrm{ft}$ utilizing the most commonly used approximate manual design procedures and the exact design procedure. The following approximate procedures are discussed in this example:

1. Bresler Reciprocal Load Method
2. Bresler Load Contour Method

## 3. PCA Load Contour Method

Figure 1 shows the reinforced concrete square column cross section under consideration. The calculated approximate values are compared with the exact hand calculated values and automated results obtained from the spColumn engineering software program from StructurePoint formerly the PCA engineering Software Group. The steps to develop the three-dimensional failure surface (3D interaction diagram) using spColumn is shown in detail as well.


Figure 1 - Reinforced Concrete Column Cross-Section

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## Code

Building Code Requirements for Structural Concrete (ACI 318-11) and Commentary (ACI 318R-11)
Building Code Requirements for Structural Concrete (ACI 318-14) and Commentary (ACI 318R-14)

Building Code Requirements for Structural Concrete (ACI 318-19) and Commentary (ACI 318R-19)

## References

- Notes on ACI 318-11 Building Code Requirements for Structural Concrete, Twelfth Edition, 2013 Portland Cement Association, Example 7.8 (this reference will be referred as "PCA Notes" in the rest of this document)
- spColumn Engineering Software Program Manual v10.00, STRUCTUREPOINT, 2021
- "Biaxial Bending Interaction Diagrams for Square Reinforced Concrete Column Design (ACI 318-19)" Design Example, STRUCTUREPOINT, 2022
- "Biaxial Bending Interaction Diagrams for Rectangular Reinforced Concrete Column Design (ACI 318-19)" Design Example, STRUCTUREPOINT, 2022
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- Capacity of Reinforced Rectangular Columns Subject to Biaxial Bending, Publication EB011D, Portland Cement Association, Skokie, IL, 1966.
- Biaxial and Uniaxial Capacity of Rectangular Columns, Publication EB031D, Portland Cement Association, Skokie, IL, 1967.


## Design Data

$f_{c}{ }^{\prime}=5000 \mathrm{psi}$
$f_{y}=60000 \mathrm{psi}$
Cover $=2.0 \mathrm{in}$.
Reinforcement $=4$ \#11
Column dimensions and reinforcement locations are shown in Figure 2.


Figure 2 - Reinforced Concrete Column Cross-Section and Reinforcement Locations

## Solution - Manual Design Procedure

The nominal strength of a section under biaxial bending and compression is a function of three variables $\mathrm{P}_{\mathrm{n}}, \mathrm{M}_{\mathrm{nx}}$ and $M_{n y}$ which may be expressed in terms of an axial load acting at eccentricities $e_{x}=M_{n y} / P_{n}$ and $e_{y}=M_{n x} / P_{n}$ as shown in Figure 3. A failure surface may be described as a surface produced by plotting the failure load $\mathrm{P}_{\mathrm{n}}$ as a function of its eccentricities $e_{x}$ and $e_{y}$, or of its associated bending moments $M_{n y}$ and $M_{n x}$. Three types of failure surfaces have been defined. The basic surface $S_{1}$ is defined by a function which is dependent upon the variables $P_{n}, e_{x}$ and $e_{y}$, as shown in Figure 4(a). a reciprocal surface can be derived from $S_{1}$ in which the reciprocal of the nominal axial load $P_{n}$ is employed to produce the surface $S_{2}\left(1 / \mathrm{P}_{\mathrm{n}}, \mathrm{e}_{\mathrm{x}}, \mathrm{e}_{\mathrm{y}}\right)$ as illustrated in Figure 4(b). the third type of failure surface, shown in Figure 4(c), is obtained by relating the nominal axial load $P_{n}$ to the moments $M_{n x}$ and $M_{n y}$ to produce surface $S_{3}\left(P_{n}\right.$, $M_{n x}, M_{n y}$ ). Failure surface $S_{3}$ is the three-dimensional extension of the uniaxial interaction diagram previously described. A number of investigators have made approximations for both the $S_{2}$ and $S_{3}$ failure surfaces for use in design and analysis. An explanation of these methods used in current practice is given below.


Reinforcing bars not shown

Figure 3 - Notation for Biaxial Loading


Figure 4 - Failure Surfaces

The following outlines the approximate manual design procedure proposed by the reference to estimate the capacity of column or wall sections subjected to combined axial load and biaxial bending moments. The procedure is comprised of four key steps where the fourth step covers the approximate biaxial methods.

STEP A) Choose the value of $\beta$ at 0.65 or use Figures $\mathbf{7 - 1 5}$ and $\mathbf{7 - 1 6}$ in Chapter 7 in PCA Notes to make an estimate.


Figure 5 -Sample of Figure 7-15 (Biaxial Design Constants - 4 Bar Arrangement)
For this example, assume $\beta=0.65$ and compression-controlled behavior
$\therefore \phi=0.65$
PCA Notes (Chapter 7 Eq. 17)

The minimum required strengths equal to:

$$
\begin{aligned}
& P_{n_{-} \text {req }}=\frac{P_{u}}{\phi}=\frac{1200}{0.65}=1846 \mathrm{kips} \\
& M_{n x_{-} \text {req }}=\frac{M_{u x}}{\phi}=\frac{300}{0.65}=461.5 \mathrm{kip}-\mathrm{ft} \\
& M_{n y_{-} \text {req }} \frac{M_{u y}}{\phi}=\frac{125}{0.65}=192.3 \mathrm{kip-ft}
\end{aligned}
$$

STEP B) If $M_{n y \_ \text {req }} / M_{n x \_ \text {req }}$ is greater than $\mathrm{b} / \mathrm{h}$, use Eq. (17) in Chapter 7 in PCA Notes to calculate an approximate equivalent uniaxial moment strength $\mathrm{M}_{\text {noy_req. }}$. If $\mathrm{M}_{\mathrm{ny} \_ \text {req }} / \mathrm{M}_{\mathrm{nx} \_ \text {req }}$ is less than $\mathrm{b} / \mathrm{h}$, use Eq. (20) in Chapter 7 in PCA Notes to calculate an approximate equivalent uniaxial moment strength $\mathrm{M}_{\text {nox_req }}$.

PCA Notes (Chapter 7 Eq. 17)

$$
\begin{aligned}
& M_{n x_{-} \text {req }} \times \frac{b}{h} \times\left(\frac{1-\beta}{\beta}\right)+M_{n y_{-} \text {req }} \approx M_{n o y_{-} \text {req }} \\
& M_{n x_{-} \text {req }}+M_{n y_{-} \text {req }} \times \frac{h}{b} \times\left(\frac{1-\beta}{\beta}\right) \approx M_{n o x_{-} r e q} \\
& \frac{M_{n y_{-} \text {req }}}{M_{n x_{-} \text {req }}}=\frac{192.3}{461.5}=0.42<\frac{b}{h}=\frac{24}{24}=1.00 \rightarrow \text { Use Eq. } 20 \\
& M_{n o x_{-} \text {req }} \approx 461.5+192.3 \times \frac{24}{24} \times\left(\frac{1-0.65}{0.65}\right)=565.1 \mathrm{kip}-\mathrm{ft}
\end{aligned}
$$

STEP C) Design the section using any suitable approximate or exact method for uniaxial bending with axial load to provide an axial load strength $\mathrm{P}_{\mathrm{n}_{-} \text {req }}$ and an equivalent uniaxial moment strength $\mathrm{M}_{\text {noy_req }}$ or $\mathrm{M}_{\text {nox_req }}$. In this example, the exact calculations of column strength for uniaxial bending with axial load is used. Complete details about this procedure is provided in "Interaction Diagram - Tied Reinforced Concrete Column Design Strength (ACI 318-19)" design example. Using the same procedure explained in the design example (Summarized in Figure 6), a 24 in . square column with 4 \#11 bars provides the following capacities:
$\mathrm{P}_{\mathrm{n}}=1846=\mathrm{P}_{\mathrm{n} \_ \text {req }}$
$\mathrm{M}_{\mathrm{nox}}=682.8 \mathrm{kip}-\mathrm{ft}>\mathrm{M}_{\text {nox_req }}=565.1 \mathrm{kip}-\mathrm{ft}$
$\therefore$ The section is adequate with this reinforcement for $\left(\mathrm{P}_{\mathrm{n}_{-} \text {req }}, \mathrm{M}_{\text {nox_req }}\right)$.


Figure 6 - Column Section Capacity Interaction Diagram (spColumn)

STEP D) Verify the suitability of the chosen section by any one of the following four methods:

1. Bresler Reciprocal Load Method.
2. Bresler Load Contour Method.
3. PCA Load Contour Method.
4. Exact Biaxial Bending-Axial Interaction Method.

## 1. Bresler Reciprocal Load Method

### 1.1. Theory

This method approximates the ordinate $1 / \mathrm{P}_{\mathrm{n}}$ on the surface $\mathrm{S}_{2}\left(1 / \mathrm{P}_{\mathrm{n}}, \mathrm{e}_{\mathrm{x}}, \mathrm{e}_{\mathrm{y}}\right)$ by a corresponding ordinate $1 / \mathrm{P}_{\mathrm{n}}{ }^{\prime}$ on the plane $S_{2}^{\prime}\left(1 / P_{n}^{\prime}, e_{x}, e_{y}\right)$, which is defined by the characteristic points $A, B$ and $C$, as indicated in Figure 7. For any particular cross-section, the value $\mathrm{P}_{\mathrm{o}}$ (corresponding to point C ) is the load strength under pure axial compression; $\mathrm{P}_{\mathrm{ox}}$ (corresponding to point B ) and $\mathrm{P}_{\mathrm{oy}}$ (corresponding to point A ) are the load strengths under uniaxial eccentricities $e_{y}$ and $e_{x}$, respectively. Each point on the true surface is approximated by a different plane; therefore, the entire surface is approximated using an infinite number of planes.


Figure 7 - Reciprocal Load Method
The general expression for axial load strength for any values of $e_{x}$ and $e_{y}$ is as follows:

$$
P_{n}=\frac{1}{\frac{1}{P_{o x}}+\frac{1}{P_{o y}}-\frac{1}{P_{o}}}
$$

where

$$
\begin{aligned}
& \mathrm{P}_{\mathrm{ox}}=\text { Maximum uniaxial load strength of the column with a moment of } \mathrm{M}_{\mathrm{nx} \_ \text {req }}=\mathrm{P}_{\mathrm{n} \_ \text {_req } \times \mathrm{e}_{\mathrm{y}}} \\
& \mathrm{P}_{\mathrm{oy}}=\text { Maximum uniaxial load strength of the column with a moment of } \mathrm{M}_{\mathrm{ny} \mathrm{\_r}}=\mathrm{P}_{\mathrm{n}_{\_} \text {req }} \times \mathrm{e}_{\mathrm{x}} \\
& \mathrm{P}_{\mathrm{o}}=\text { Maximum axial load strength with no applied moments }
\end{aligned}
$$

This equation is simple in form and the variables are easily determined. Axial load strengths $\mathrm{P}_{\mathrm{o}}, \mathrm{P}_{\mathrm{ox}}$, and $\mathrm{P}_{\mathrm{oy}}$ are determined using the method presented in "Interaction Diagram - Tied Reinforced Concrete Column Design

Strength (ACI 318-19)" design example. Experimental results have shown the above equation to be reasonably accurate when flexure does not govern design. The equation should only be used when:

$$
P_{n} \geq 0.1 \times f_{c}^{\prime} \times A_{g}
$$

PCA Notes (Chapter 7 Eq. 8)

### 1.2. Calculations

Check if flexure does not govern design:

$$
P_{n} \geq 0.1 \times f_{c}^{\prime} \times A_{g}
$$

PCA Notes (Chapter 7 Eq. 8)

$$
\begin{equation*}
P_{n}=1846 \text { kip } \geq 0.1 \times f_{c}^{\prime} \times A_{g}=0.1 \times 5000 \times(24 \times 24)=288 \text { kip } \tag{O.K.}
\end{equation*}
$$

$\mathrm{P}_{\mathrm{o}}, \mathrm{P}_{\mathrm{ox}}$, and $\mathrm{P}_{\mathrm{oy}}$ need to be determined to use this method:

$$
\begin{aligned}
& P_{o}=0.85 \times f_{c}^{\prime} \times\left(A_{g}-A_{s t}\right)+f_{y} \times A_{s t} \\
& P_{o}=0.85 \times 5000 \times(24 \times 24-4 \times 1.56)+60000 \times(4 \times 1.56)=2796 \mathrm{kip}
\end{aligned}
$$

$P_{\mathrm{ox}}$ is the uniaxial load strength when only $\mathrm{M}_{\mathrm{nx} \_ \text {req }}$ acts on the column. As shown in Figure 8, for $\mathrm{M}_{\mathrm{nx} \_ \text {req }}=461.5$ kip-ft, $\mathrm{P}_{\mathrm{ox}}$ equals to 2241 kip.


PM at 0.0 [deg]

## 

Similarly, $\mathrm{P}_{\text {oy }}$ is the uniaxial load strength when only $\mathrm{M}_{\mathrm{ny} \text { _req }}$ acts on the column. As shown in Figure 9, for $\mathrm{M}_{\text {ny_req }}=192.3 \mathrm{kip}-\mathrm{ft}, \mathrm{P}_{\text {oy }}$ equals to 2579.5 kip .
$\qquad$


PM at 0.0 [deg]
Figure 9 - Column Uniaxial Interaction Diagram ( $\mathrm{P}_{\text {noy }}$ and $\mathrm{M}_{\text {ny_req }}$ )
Note that both $\mathrm{P}_{\text {ox }}(2241 \mathrm{kip})$ and $\mathrm{P}_{\text {oy }}(2579.5 \mathrm{kip})$ are greater than the balanced axial force ( 1049.2 kip ), so that the section is compression-controlled.

Using the values above, the axial load strength equals to:

$$
\begin{align*}
& P_{n}=\frac{1}{\frac{1}{P_{o x}}+\frac{1}{P_{o y}}-\frac{1}{P_{o}}} \\
& P_{n}=\frac{1}{\frac{1}{2241.0}+\frac{1}{2579.5}-\frac{1}{2796.0}}=2100.0 \mathrm{kip}>P_{n}=1846 \mathrm{kip} \tag{O.K.}
\end{align*}
$$

PCA Notes (Chapter 7 Eq. 7)

## 2. Bresler Load Contour Method

### 2.1. Theory

In this method, the surface $\mathrm{S}_{3}\left(\mathrm{P}_{\mathrm{n}}, \mathrm{M}_{\mathrm{nx}}, \mathrm{M}_{\mathrm{ny}}\right)$ is approximated by a family of curves corresponding to constant values of $P_{n}$. These curves, as illustrated in Figure 10, may be regarded as "load contours".


Figure 10 - Bresler Load Contours for Constant $\mathrm{P}_{\underline{n}}$ on Failure Surface $\underline{S}_{\underline{3}}$
The general expression for these curves can be approximated by a nondimensional interaction equation of the form:

$$
\left(\frac{M_{n x}}{M_{n o x}}\right)^{\alpha}+\left(\frac{M_{n y}}{M_{n o y}}\right)^{\beta}=1.0
$$

PCA Notes (Chapter 7 Eq. 9)
where

$$
\begin{aligned}
& \mathrm{M}_{\mathrm{nx}}=\text { The nominal biaxial moment strength in the direction of } \mathrm{x} \text { axis } \\
& \mathrm{M}_{\mathrm{ny}}=\text { The nominal biaxial moment strength in the direction of } \mathrm{y} \text { axis } \\
& \mathrm{M}_{\mathrm{nox}}=\text { The nominal uniaxial moment strength about the } \mathrm{x} \text {-axis } \\
& \mathrm{M}_{\mathrm{noy}}=\text { The nominal uniaxial moment strength about the } \mathrm{y} \text {-axis }
\end{aligned}
$$

Note that ( $\mathrm{M}_{\mathrm{nx}}$ and $\mathrm{M}_{\mathrm{ny}}$ ) are the vectorial equivalent of the nominal uniaxial moment $\mathrm{M}_{\mathrm{n}}$. The values of the exponents $\alpha$ and $\beta$ are a function of the amount, distribution and location of reinforcement, the dimensions of
the column, and the strength and elastic properties of the steel and concrete. Bresler indicates that it is reasonably accurate to assume that $\alpha=\beta$; therefore, the previous equation becomes (shown graphically in Figure 11):

$$
\left(\frac{M_{n x}}{M_{n o x}}\right)^{\alpha}+\left(\frac{M_{n y}}{M_{n o y}}\right)^{\alpha}=1.0
$$

PCA Notes (Chapter 7 Eq. 10)


Figure 11 - Interaction Curves for Bresler Load Contour Method
When using the previous equation or figure, it is still necessary to determine the $\alpha$ value for the cross-section being designed. Bresler indicated that, typically, $\alpha$ varied from 1.15 to 1.55 , with a value of 1.5 being reasonably accurate for most square and rectangular sections having uniformly distributed reinforcement.

With $\alpha$ set at unity, the interaction equation becomes linear:

$$
\frac{M_{n x}}{M_{n o x}}+\frac{M_{n y}}{M_{n o y}}=1.0
$$

PCA Notes (Chapter 7 Eq. 11)

The previous equation would always yield conservative results since it underestimates the column capacity, especially for high axial loads or low percentages of reinforcement. It should only be used when:

$$
P_{n}<0.1 \times f_{c}^{\prime} \times A_{g}
$$

PCA Notes (Chapter 7 Eq. 12)

### 2.2. Calculations

The reference conservatively selected $\alpha=1.0$ due to a lack of available data. Check if flexure does not govern design:

$$
\begin{aligned}
& P_{n}<0.1 \times f_{c}^{\prime} \times A_{g} \\
& P_{n}=1846 \mathrm{kip}>0.1 \times f_{c}^{\prime} \times A_{g}=0.1 \times 5000 \times(24 \times 24)=288 \mathrm{kip}
\end{aligned}
$$

PCA Notes (Chapter 7 Eq. 12)
(No Good)

Although $P_{n}>0.1 \times f_{c}^{\prime} \times A_{g}$, the reference decided to carry out the necessary calculations for illustration purposes.

Since the section is symmetrical:

$$
M_{n o x}=M_{n o y}=682.8 \text { kip-ft (Figure 6) }
$$

Using the interaction equation becomes linear:

$$
\begin{align*}
& \frac{M_{n x}}{M_{n o x}}+\frac{M_{n y}}{M_{n o y}}=1.0 \\
& \frac{M_{n x-r e q}}{M_{n o x}}+\frac{M_{n y-r e q}}{M_{n o y}}=\frac{461.5}{682.8}+\frac{192.3}{682.8}=0.96<1.0=\frac{M_{n x}}{M_{n o x}}+\frac{M_{n y}}{M_{\text {noy }}} \tag{O.K.}
\end{align*}
$$

## 3. PCA Load Contour Method

### 3.1. Theory

The PCA approach described below was developed as an extension of the Bresler Load Contour Method. The Bresler interaction equation was chosen as the most viable method in terms of accuracy, practicality, and simplification potential.

A typical Bresler load contour for a certain $P_{n}$ is shown in the following Figure 12. In the PCA method, point B is defined such that the nominal biaxial moment strengths $\mathrm{M}_{\mathrm{nx}}$ and $\mathrm{M}_{\mathrm{ny}}$ at this point are in the same ratio as the uniaxial moment strengths $M_{n o x}$ and $M_{n o y}$. Therefore, at point $B\left(M_{n x} / M_{n y}=M_{n o x} / M_{n o y}\right)$.


Figure 12 - Load Contour of Failure Surface $\mathrm{S}_{\underline{3}}$ along Plane of Constant $\mathrm{P}_{\underline{n}}$
When the load contour of Figure 12 is nondimensionalized, it takes the form shown in Figure 13, and the point $B$ will have x and y coordinates of $\beta$. When the bending resistance is plotted in terms of the dimensionless parameters $P_{n} / P_{0}, M_{n x} / M_{n o x}, M_{n y} / M_{n o y}$ (the latter two designated as the relative moments), the generated failure surface $\mathrm{S} 4\left(\mathrm{P}_{\mathrm{n}} / \mathrm{P}_{\mathrm{o}}, \mathrm{M}_{\mathrm{nx}} / \mathrm{M}_{\mathrm{nox}}, \mathrm{M}_{\mathrm{ny}} / \mathrm{M}_{\mathrm{noy}}\right)$ assumes the typical shape shown in Figure 14 . The advantage of expressing the behavior in relative terms is that the contours of the surface (Figure 13) - i.e., the intersection formed by planes of constant $\mathrm{P}_{\mathrm{n}} / \mathrm{P}_{\mathrm{o}}$ and the surface - can be considered for design purposes to be symmetrical about the vertical plane bisecting the two coordinate planes. Even for sections that are rectangular or have unequal reinforcement on the two adjacent faces, this approximation yields values sufficiently accurate for design.


Figure 13 - Nondimensional Load Contour at Constant $\mathrm{P}_{\underline{n}}$


Figure 14 - Nondimensional Load Contour at Constant $\mathrm{P}_{\underline{n}}$
The reference obtained the relationship between $\alpha$ from Eq. (10) and $\beta$ by substituting the coordinates of point B from Figure 11 into Eq. (10), and solving for $\alpha$ in terms of $\beta$. This yields:

$$
\alpha=\frac{\log 0.5}{\log \beta}
$$

Thus, Eq. (10) may be written as:

$$
\left(\frac{M_{n x}}{M_{n o x}}\right)^{\left(\frac{\log 0.5}{\log \beta}\right)}+\left(\frac{M_{n y}}{M_{n o y}}\right)^{\left(\frac{\log 0.5}{\log \beta}\right)}=1.0
$$

PCA Notes (Chapter 7 Eq. 14)

For design convenience, a plot of the curves generated by Eq. (14) for nine values of $\beta$ are given in Figure 15. Note that when $\beta=0.5$, its lower limit, Eq. (14) is a straight line joining the points at which the relative moments equal 1.0 along the coordinate planes. When $\beta=1.0$, its upper limit, Eq. (14) is two lines, each of which is parallel to one of the coordinate planes.


Figure 15 - Biaxial Moment Strength Relationship
$\beta$ is dependent primarily on the ratio $P_{n} / P_{o}$ and to a lesser, though still significant extent, on the bar arrangement, the reinforcement index $\omega$ and the strength of the reinforcement as shown in Figure 16.


Figure 16 - Biaxial Design Constants
Rapid and easy convergence to a satisfactory section can be achieved by approximating the curves in Figure 15 by two straight lines intersecting at the $45^{\circ}$ line, as shown in Figure 17.


Figure 17 - Bilinear Approximation of Nondimensionalized Load Contour

$$
\frac{M_{n x}}{M_{n o x}} \times\left(\frac{1-\beta}{\beta}\right)+\frac{M_{n y}}{M_{n o y}}=1.0 \text { for } \frac{M_{n y}}{M_{n x}}>\frac{M_{n o y}}{M_{n o x}}
$$

which can be restated for design convenience as follows:

$$
M_{n x} \times\left(\frac{M_{n o y}}{M_{n o x}}\right) \times\left(\frac{1-\beta}{\beta}\right)+M_{n y}=M_{n o y}
$$

PCA Notes (Chapter 7 Eq. 16)

For rectangular sections with reinforcement equally distributed on all faces, Eq. (16) can be approximated by:

$$
M_{n x} \times\left(\frac{b}{h}\right) \times\left(\frac{1-\beta}{\beta}\right)+M_{n y} \approx M_{n o y}
$$

PCA Notes (Chapter 7 Eq. 17)

The equation of the lower line of Figure 17 is:

$$
\begin{aligned}
& \frac{M_{n x}}{M_{n o x}}+\frac{M_{n y}}{M_{n o y}} \times\left(\frac{1-\beta}{\beta}\right)=1.0 \text { for } \frac{M_{n y}}{M_{n x}}<\frac{M_{n o y}}{M_{n o x}} \\
& M_{n x}+M_{n y} \times\left(\frac{M_{n o x}}{M_{n o y}}\right) \times\left(\frac{1-\beta}{\beta}\right)=M_{n o x}
\end{aligned}
$$

PCA Notes (Chapter 7 Eq. 19)

For rectangular sections with reinforcement equally distributed on all faces,

$$
M_{n x}+M_{n y} \times\left(\frac{h}{b}\right) \times\left(\frac{1-\beta}{\beta}\right) \approx M_{n o x}
$$

PCA Notes (Chapter 7 Eq. 20)

In design Eqs. (17) and (20), the ratio $b / h$ or $h / b$ must be chosen and the value of $\beta$ must be assumed. For lightly loaded columns, $\beta$ will generally vary from 0.55 to about 0.70 . Hence, a value of 0.65 for $\beta$ is generally a good initial choice in a biaxial bending analysis.

### 3.2. Calculations

$\mathrm{P}_{0}, \mathrm{M}_{\mathrm{nox}}, \mathrm{M}_{\text {noy }}$ and $\beta$ need to be determined to use this method:

$$
\begin{aligned}
& P_{o}=0.85 \times f_{c}^{\prime} \times\left(A_{g}-A_{s t}\right)+f_{y} \times A_{s t} \\
& P_{o}=0.85 \times 5000 \times(24 \times 24-4 \times 1.56)+60000 \times(4 \times 1.56)=2796 \mathrm{kip}
\end{aligned}
$$

Since the section is symmetrical:

$$
M_{\text {nox }}=M_{n o y}=682.8 \text { kip-ft (Figure 6) }
$$

$\beta$ can be determined using $\mathrm{P}_{\mathrm{o}}, \rho_{\mathrm{g}}$ and the biaxial design constants figure:

$$
\begin{aligned}
& \frac{P_{n}}{P_{o}}=\frac{1846}{2796}=0.66 \\
& \omega=\frac{\rho_{g} \times f_{y}}{f_{c}^{\prime}}=\frac{0.011 \times 60000}{5000}=0.13, \text { where } \rho_{g}=\frac{A_{s t}}{b \times h}=\frac{4 \times 1.56}{24 \times 24}=0.011 \\
& \beta=0.66 \text { (Figure 18): }
\end{aligned}
$$

4 bar arrangement


Figure 18 - Biaxial Design Constants (4 Bar Arrangement)

$$
\begin{align*}
& \left(\frac{M_{n x_{-} \text {req }}}{M_{n o x}}\right)^{\left(\frac{\log 0.5}{\log \beta}\right)}+\left(\frac{M_{n y-r e q}}{M_{n o y}}\right)^{\left(\frac{\log 0.5}{\log \beta}\right)}<\left(\frac{M_{n x}}{M_{n o x}}\right)^{\left(\frac{\log 0.5}{\log \beta}\right)}+\left(\frac{M_{n y}}{M_{n o y}}\right)^{\left(\frac{\log 0.5}{\log \beta}\right)} \\
& \left(\frac{461.5}{682.8}\right)^{\left(\frac{\log 0.5}{\log 0.66}\right)}+\left(\frac{192.3}{682.8}\right)^{\left(\frac{\log 0.5}{(\log 0.66}\right)}=0.52+0.12=0.64<1.0 \tag{O.K.}
\end{align*}
$$

Using bilinear approximation (Eqs. 15 and 18):

$$
\frac{M_{n y \_ \text {req }}}{M_{n x_{-} \text {req }}}=\frac{192.3}{461.5}=0.42<\frac{M_{\text {noy }}}{M_{\text {nox }}}=\frac{682.8}{682.8}=1.00
$$

$\therefore$ Eq. 18 should be used

$$
\begin{aligned}
& \frac{M_{n x}}{M_{n o x}}+\frac{M_{n y}}{M_{n o y}} \times\left(\frac{1-\beta}{\beta}\right)=1.0 \text { for } \frac{M_{n y}}{M_{n x}}<\frac{M_{n o y}}{M_{n o x}} \\
& \frac{M_{n x \_r e q}}{M_{n o x}}+\frac{M_{n y \_r e q}}{M_{n o y}} \times\left(\frac{1-\beta}{\beta}\right)<\frac{M_{n x}}{M_{n o x}}+\frac{M_{n y}}{M_{n o y}} \times\left(\frac{1-\beta}{\beta}\right) \\
& \frac{461.5}{682.8}+\frac{192.3}{682.8} \times\left(\frac{1-0.66}{0.66}\right)=0.676+0.145=0.82<1.0
\end{aligned}
$$

PCA Notes (Chapter 7 Eq. 18)
(O.K.)

## 4. Exact Biaxial Bending-Axial Interaction Method

### 4.1. Theory

In a reinforced concrete column, the exact calculations of section capacity (axial force biaxial moment strengths) involves a trial-and-error process for calculating the neutral axis depth and angle $\alpha$. The steps to calculate biaxial flexural strength of a rectangular reinforced concrete column for a given nominal axial strength are as follows:

1. Assuming a value for the angle of the neutral axis ( $\alpha$ ) and the neutral axis depth (c) and calculating the strain values in each reinforcement layer
2. Calculating the forces values in the concrete $\left(C_{c}\right)$ and reinforcement layers ( $F_{s i}$ )
3. Calculating $\mathrm{P}_{\mathrm{n}}, \mathrm{M}_{\mathrm{nx}}$ and $\mathrm{M}_{\mathrm{ny}}$ using the following equations

$$
\begin{aligned}
& P_{n}=C_{c}+\sum F_{s} \\
& M_{n x}=C_{c} \times\left(\frac{h}{2}-\bar{y}_{c}\right)+\sum_{i=1}^{n=10}\left(F_{s i} \times\left(\frac{h}{2}-y_{i}\right)\right) \\
& M_{n y}=C_{c} \times\left(\frac{b}{2}-\bar{x}_{c}\right)+\sum_{i=1}^{n=10}\left(F_{s i} \times\left(\frac{b}{2}-x_{i}\right)\right)
\end{aligned}
$$

The procedure above should be repeated until the calculated $P_{n}$ is equal to the given $P_{n}$.
The difficulty associated with the determination of the strength of reinforced columns subjected to combined axial load and biaxial bending is primarily an arithmetic one. The bending resistance of an axially loaded column about a particular skewed axis is determined through iterations involving simple but lengthy calculations. These extensive calculations are compounded when optimization of the reinforcement or cross-section is sought.

### 4.2. Calculations

Figure 19 summarizes the exact calculations procedure to determine column section capacity subjected to axial load and biaxial bending moments. This procedure is illustrated in detail in several design examples published by StructurePoint:

- Biaxial Bending Interaction Diagrams for Square Reinforced Concrete Column Design (ACI 318-19)
- Biaxial Bending Interaction Diagrams for Rectangular Reinforced Concrete Column Design (ACI 318-19)
- Biaxial Bending Interaction Diagrams for C-Shaped Concrete Core Wall Design (ACI 318-19)
- Biaxial Bending Interaction Diagrams for Spiral Reinforced Circular Concrete Column Design (ACI 31819)


Figure 19 - Column Biaxial Bending - Axial Interaction Exact Method.

Following the procedure described above:

$$
\begin{align*}
& c=25.11 \mathrm{in} . \\
& \alpha=25.24^{\circ} \\
& P_{n}=C_{c}+\sum F_{s}=1846 \mathrm{kip}=P_{n_{-} \text {req }}  \tag{O.K.}\\
& M_{n x}=C_{c} \times\left(\frac{h}{2}-\bar{y}_{c}\right)+\sum_{i=1}^{n}\left(F_{s i} \times\left(\frac{h}{2}-y_{i}\right)\right)=608 \mathrm{kip}-\mathrm{ft}>M_{n x_{-} \text {req }}=461.5 \mathrm{kip}-\mathrm{ft}  \tag{O.K.}\\
& M_{n y}=C_{c} \times\left(\frac{b}{2}-\bar{x}_{c}\right)+\sum_{i=1}^{n}\left(F_{s i} \times\left(\frac{b}{2}-x_{i}\right)\right)=245.4 \mathrm{kip}-\mathrm{ft}>M_{n y_{-} \text {req }}=192.3 \mathrm{kip}-\mathrm{ft} \tag{O.K.}
\end{align*}
$$

## 5. Column Biaxial Bending Interaction Diagram - spColumn Software

spColumn is a StructurePoint software program that performs the analysis and design of reinforced concrete sections subjected to axial force combined with uniaxial or biaxial bending. Using the provisions of the Strength Design Method and Unified Design Provisions, slenderness considerations are used for moment magnification due to second order effect (P-Delta) for sway and non-sway frames.

For biaxial runs, the values of maximum compressive axial load capacity and maximum tensile load capacity are computed. These two values set the range within which the moment capacities are computed for a predetermined number of axial load values. For each level of axial load, the section is rotated in 10-degree increments from 0 degrees to 360 degrees and the $\mathrm{M}_{\mathrm{x}}$ and $\mathrm{M}_{\mathrm{y}}$ moment capacities are computed. Thus, for each level of axial load, an $M_{x}-M_{y}$ contour is developed. Repeating this for the entire range of axial loads, the three-dimensional failure surface is computed. A three-dimensional visualization of the resulting entire nominal and factored failure surface is provided to support enhanced understanding of the section capacity.

The "biaxial" feature allows the user to investigate the P-M interaction diagrams, the $\mathrm{M}_{\mathrm{x}}-\mathrm{M}_{\mathrm{y}}$ moment contour plots, as well as the 3D failure surface for even the most irregular column and shear wall sections quickly, simply, and accurately.

For this column section, we ran in investigation mode with "biaxial" option for "Run Axis" using the ACI 31819.

Structure Point


Figure 20 - spColumn Interface


Figure 21 - spColumn Model Editor


Figure 22 - Defining Loads / Modes (spColumn)


Figure 23 - Column Section $\mathrm{M}_{\underline{n x}}-\mathrm{M}_{\underline{n} \underline{2}}$ Contour at $\mathrm{P}_{\underline{n}}=1846$ kip (spColumn)


[^0]STRUCTUREPOINT - spColumn v10.00 (TM) Page | 2
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## 1. General Information

| File Name | E:IStructurePointlspColu...IPCA <br> Notes Ex 7.8.colx |
| :--- | :--- |
| Project | PCA Notes Example |
| Column | Ex. 7.8 |
| Engineer | SP |
| Code | ACI 318-19 |
| Bar Set | ASTM A615 |
| Units | English |
| Run Option | Investigation |
| Run Axis | Biaxial |
| Slenderness | Not Considered |
| Column Type | Structural |
| Capacity Method | Moment capacity |

## 2. Material Properties

### 2.1. Concrete

| Type | Standard |
| :--- | :--- |
| $f_{c}$ | 5 ksi |
| $E_{c}$ | 4030.51 ksi |
| $f_{c}$ | 4.25 ksi |
| $\varepsilon_{u}$ | $0.003 \mathrm{in} / \mathrm{in}$ |
| $\beta_{1}$ | 0.8 |

2.2. Steel

| Type | Standard |
| :--- | :--- |
| $\mathrm{f}_{\mathrm{y}}$ | 60 ksi |
| $\mathrm{E}_{\mathrm{s}}$ | 29000 ksi |
| $\varepsilon_{\mathrm{ty}}$ | $0.00206897 \mathrm{in} / \mathrm{in}$ |

## 3. Section

3.1. Shape and Properties

| Type | Irregular |
| :--- | :---: |
| $A_{g}$ | $576 \mathrm{in}^{2}$ |
| $I_{x}$ | $27648 \mathrm{in}^{4}$ |
| $I_{y}$ | $27648 \mathrm{in}^{4}$ |
| $r_{x}$ | 6.9282 in |
| $r_{y}$ | 6.9282 in |
| $X_{0}$ | 0 in |
| $Y_{0}$ | 0 in |

### 3.2. Section Figure



Figure 1: Column section
3.3. Solids
3.3.1. S1

| Points | X <br> in | $\mathbf{Y}$ <br> in | Points | $\mathbf{X}$ <br> in | $\mathbf{Y}$ <br> in | Points | $\mathbf{X}$ <br> in | $\mathbf{Y}$ <br> in |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| 1 | -12.0 | 12.0 | 2 | -12.0 | -12.0 | 3 | 12.0 | -12.0 |
| 4 | 12.0 | 12.0 |  |  |  |  |  |  |

## 4. Reinforcement

4.1. Bar Set: ASTM A615

| Bar | Diameter <br> in | Area <br> in $^{2}$ | Bar | Diameter <br> in | Area <br> in $^{2}$ | Bar | Diameter <br> in | Area <br> in $^{2}$ |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| $\# 3$ | 0.38 | 0.11 | $\# 4$ | 0.50 | 0.20 | $\# 5$ | 0.63 | 0.31 |
| $\# 6$ | 0.75 | 0.44 | $\# 7$ | 0.88 | 0.60 | $\# 8$ | 1.00 | 0.79 |
| $\# 9$ | 1.13 | 1.00 | $\# 10$ | 1.27 | 1.27 | $\# 11$ | 1.41 | 1.56 |
| $\# 14$ | 1.69 | 2.25 | $\# 18$ | 2.26 | 4.00 |  |  |  |

### 4.2. Confinement and Factors

| Confinement type | Other |
| :--- | :---: |
| For \#10 bars or less | \#3 ties |
| For larger bars | \#4 ties |
| Capacity Reduction Factors |  |
| Axial compression, (a) | 1 |
| Tension controlled $\phi$, (b) | 1 |
| Compression controlled $\phi$, (c) | 1 |

Capacity reduction factors set to 1.0 in spColumn input for illustration purposes

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E:IStructurePointlspColumnl12 Biaxial Manual DesignlPCA Notes Ex 7.8.colx

### 4.3. Arrangement

| Pattern | Irregular |
| :--- | :---: |
| Bar layout | --- |
| Cover to | --- |
| Clear cover | --- |
| Bars |  |
|  | $6.24 \mathrm{in}^{2}$ |
| Total steel area, $A_{s}$ | $1.08 \%$ |
| Rho | 18.59 in |
| Minimum clear spacing |  |

### 4.4. Bars Provided

| Area |  |  |  |  |  |  |  |  |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| in $^{2}$ | X <br> in | $\mathbf{Y}$ <br> in | Area <br> in $^{2}$ | $\mathbf{X}$ <br> in | $\mathbf{Y}$ <br> in | Area <br> in $^{2}$ | $\mathbf{X}$ <br> in | $\mathbf{Y}$ <br> in |
| 1.56 | -10.0 | -10.0 | 1.56 | -10.0 | 10.0 | 1.56 | 10.0 | -10.0 |
| 1.56 | 10.0 | 10.0 |  |  |  |  |  |  |

5. Control Points

| About Point | + | X-Moment k-ft | Y-Moment NA Depth |  | $\mathrm{d}_{\mathrm{t}} \text { Depth }$in | $\varepsilon_{\text {t }}$ | $\phi$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  | k-ft | in |  |  |  |
| X @ Max compression | 2795.9 | 0.00 | 0.00 | 70.89 | 22.00 | -0.00207 | 1.00000 |
| X @ Allowable comp. | 2795.9 | 0.00 | 0.00 | 70.89 | 22.00 | -0.00207 | 1.00000 |
| X @ $\mathrm{f}_{\mathrm{s}}=0.0$ | 1969.1 | 623.67 | 0.00 | 22.00 | 22.00 | 0.00000 | 1.00000 |
| X @ $\mathrm{f}_{\mathrm{s}}=0.5 \mathrm{ff}_{\mathrm{y}}$ | 1415.2 | 829.93 | 0.00 | 16.36 | 22.00 | 0.00103 | 1.00000 |
| X @ Balanced point | 1049.2 | 902.29 | 0.00 | 13.02 | 22.00 | 0.00207 | 1.00000 |
| X @ Tension control | 654.2 | 786.42 | 0.00 | 8.18 | 22.00 | 0.00507 | 1.00000 |
| X @ Pure bending | 0.0 | 328.59 | 0.00 | 2.11 | 22.00 | 0.02822 | 1.00000 |
| X @ Max tension | -374.4 | 0.00 | 0.00 | 0.00 | 22.00 | 9.99999 | 1.00000 |
| Y @ Max compression | 2795.9 | 0.00 | 0.00 | 70.89 | 22.00 | -0.00207 | 1.00000 |
| Y @ Allowable comp. | 2795.9 | 0.00 | 0.00 | 70.89 | 22.00 | -0.00207 | 1.00000 |
| $Y @ f_{s}=0.0$ | 1969.1 | 0.00 | 623.67 | 22.00 | 22.00 | 0.00000 | 1.00000 |
| $\mathrm{Y} @ \mathrm{f}_{\mathrm{s}}=0.5 \mathrm{f}_{\mathrm{y}}$ | 1415.2 | 0.00 | 829.93 | 16.36 | 22.00 | 0.00103 | 1.00000 |
| Y @ Balanced point | 1049.2 | 0.00 | 902.29 | 13.02 | 22.00 | 0.00207 | 1.00000 |
| Y @ Tension control | 654.2 | 0.00 | 786.42 | 8.18 | 22.00 | 0.00507 | 1.00000 |
| Y @ Pure bending | 0.0 | 0.00 | 328.59 | 2.11 | 22.00 | 0.02822 | 1.00000 |
| Y @ Max tension | -374.4 | 0.00 | 0.00 | 0.00 | 22.00 | 9.99999 | 1.00000 |
| -X @ Max compression | 2795.9 | 0.00 | 0.00 | 70.89 | 22.00 | -0.00207 | 1.00000 |
| -X @ Allowable comp. | 2795.9 | 0.00 | 0.00 | 70.89 | 22.00 | -0.00207 | 1.00000 |
| -X @ $\mathrm{f}_{\mathrm{s}}=0.0$ | 1969.1 | -623.67 | 0.00 | 22.00 | 22.00 | 0.00000 | 1.00000 |
| $-\mathrm{X} @ \mathrm{f}_{\mathrm{s}}=0.5 \mathrm{f}_{\mathrm{y}}$ | 1415.2 | -829.93 | 0.00 | 16.36 | 22.00 | 0.00103 | 1.00000 |
| -X @ Balanced point | 1049.2 | -902.29 | 0.00 | 13.02 | 22.00 | 0.00207 | 1.00000 |
| -X @ Tension control | 654.2 | -786.42 | 0.00 | 8.18 | 22.00 | 0.00507 | 1.00000 |
| -X @ Pure bending | 0.0 | -328.59 | 0.00 | 2.11 | 22.00 | 0.02822 | 1.00000 |
| -X @ Max tension | -374.4 | 0.00 | 0.00 | 0.00 | 22.00 | 9.99999 | 1.00000 |
| -Y @ Max compression | 2795.9 | 0.00 | 0.00 | 70.89 | 22.00 | -0.00207 | 1.00000 |
| -Y @ Allowable comp. | 2795.9 | 0.00 | 0.00 | 70.89 | 22.00 | -0.00207 | 1.00000 |
| $-\mathrm{Y} @ \mathrm{f}_{\mathrm{s}}=0.0$ | 1969.1 | 0.00 | -623.67 | 22.00 | 22.00 | 0.00000 | 1.00000 |
| $-\mathrm{Y} @ \mathrm{f}_{\mathrm{s}}=0.5 \mathrm{f}_{\mathrm{y}}$ | 1415.2 | 0.00 | -829.93 | 16.36 | 22.00 | 0.00103 | 1.00000 |
| -Y @ Balanced point | 1049.2 | 0.00 | -902.29 | 13.02 | 22.00 | 0.00207 | 1.00000 |
| -Y @ Tension control | 654.2 | 0.00 | -786.42 | 8.18 | 22.00 | 0.00507 | 1.00000 |
| -Y @ Pure bending | 0.0 | 0.00 | -328.59 | 2.11 | 22.00 | 0.02822 | 1.00000 |
| -Y @ Max tension | -374.4 | 0.00 | 0.00 | 0.00 | 22.00 | 9.99999 | 1.00000 |

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12:45 PM
6. Factored Loads and Moments with Corresponding Capacity Ratios NOTE: Calculations are based on "Moment Capacity" Method.

| No. | Demand |  |  | Capacity |  |  | Parameters at Capacity |  |  | Capacity Ratio |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | $\mathrm{P}_{\mathrm{u}}$ | $\mathrm{M}_{\text {ux }}$ | M ${ }_{\text {uy }}$ | $\phi \mathrm{P}_{\mathrm{n}}$ | $\phi \mathrm{M}_{\mathrm{nx}}$ | $\phi \mathrm{M}_{\mathrm{ny}}$ | NA Depth in | $\varepsilon_{t}$ | $\boldsymbol{\phi}$ |  |
|  | kip | k-ft | k-ft | kip | k-ft | k-ft |  |  |  |  |
| 1 | 1846.00 | 461.50 | 192.30 | 1846.00 | 601.97 | 250.83 | 25.11 | 0.00050 | 1.000 | 0.77 |
| 2 | 1846.00 | 565.10 | 0.00 | 1846.00 | 682.79 | 0.00 | 20.70 | 0.00019 | 1.000 | 0.83 |

7. Diagrams
7.1. PM at $\boldsymbol{\theta}=23$ [deg] [User]


## 7.2. $P M$ at $\theta=0$ [deg] [User]



## 7.3. $M M$ at $P=1846$ [kip] [User]



| General Information |  |
| :---: | :---: |
| Project | PCA Notes Example |
| Column | Ex. 7.8 |
| Engineer | SP |
| Code | ACI 318-19 |
| Bar Set | ASTM A615 |
| Units | English |
| Run Option | Investigation |
| Run Axis | Biaxial |
| Slenderness | Not Considered |
| Column Type | Structural |
| Capacity Method | Moment capacity |
| Materials |  |
| $\mathrm{f}_{\mathrm{c}}$ | 5 ksi |
| $\mathrm{E}_{\text {。 }}$ | 4030.51 ksi |
| $\mathrm{f}_{\mathrm{y}}$ | 60 ksi |
| $\mathrm{E}_{5}$ | 29000 ksi |
| Section |  |
| Type | Irregular |
| $\mathrm{A}_{0}$ | $576 \mathrm{in}^{2}$ |
| $I_{x}$ | $27648 \mathrm{in}^{4}$ |
| Iy | $27648 \mathrm{in}^{4}$ |
| Reinforcement |  |
| Pattern | Irregular |
| Bar layout | --- |
| Cover to | -- |
| Clear cover | --- |
| Bars | -- |
| Confinement type | Other |
| Total steel area, $\mathrm{As}_{5}$ | $6.24 \mathrm{in}^{2}$ |
| Rho | 1.08 \% |
| Min. clear spacing | 18.59 in |



## 6. Summary and Comparison of Results

Figure 24 shows $\mathrm{M}_{\mathrm{x}}-\mathrm{M}_{\mathrm{y}}$ contour generated by spColumn (exact solution). The approximate results obtained by Bresler Load Contour Method and PCA Load Contour Method are superimposed for comparison. There is very good agreement between the spColumn solution and the PCA curve while the Bresler curve yields straight segments given the conservative assumption of ( $\alpha=1.0$ ).
The nominal biaxial load ( $\mathrm{P}_{\mathrm{n}}, \mathrm{M}_{\mathrm{nx}}, \mathrm{M}_{\mathrm{ny}}$ ) is represented by Point 1 shown in the diagram. It is located inside all three curves indicating the column section is adequate according to all methods. Point 2 represents the equivalent uniaxial strength ( $\mathrm{P}_{\mathrm{n}}, \mathrm{M}_{\mathrm{nox}}$ ).


MM at $\mathrm{P}=1846.0$ [kip]
Figure 24 - Column Section $\mathbf{M}_{\underline{n} \underline{\underline{x}}}^{\underline{-}} \underline{\underline{n} \underline{\underline{y}}}$ Contour at $\mathrm{P}_{\underline{n}}=1846$ kip (spColumn Diagrams Module)

## 7. Conclusions \& Observations

The analysis of the reinforced concrete section performed by spColumn conforms to the provisions of the Strength Design Method and Unified Design Provisions with all conditions of strength satisfying the applicable conditions of equilibrium and strain compatibility.

In most building design calculations, such as the examples shown for flat plate or flat slab concrete floor systems, all building columns may be subjected to biaxial bending ( $\mathrm{M}_{\mathrm{x}}$ and $\mathrm{M}_{\mathrm{y}}$ ) due to lateral effects and unbalanced moments from both directions of analysis. This requires an investigation of the column $\mathrm{P}-\mathrm{M}_{\mathrm{x}}-\mathrm{M}_{\mathrm{y}}$ interaction diagram (3D failure surface) in two directions simultaneously (axial force interaction with biaxial bending).

This example shows a detailed comparison between exact hand solution, spColumn software results and the most commonly used approximate methods needed to design reinforced concrete column subjected to a combined axial force and biaxial bending moments. The approximate methods provide conservative solution in most cases and require tedious calculations to obtain the equivalent uniaxial capacities and other parameters. Additionally, performing exact solution manually to design column section subjected to a combined axial force and biaxial bending moments is tedious and challenging for engineers and the use of a computer aid can save time and eliminate errors. StucturePoint's spColumn program can, quickly, simply and accurately design and generate the three-dimensional failure surface (interaction diagram) for all commonly encountered column, beam or wall sections in addition to highly complex and irregular cross-sections.


Figure 25 - Interaction Diagram in Two Directions (Biaxial) (spColumn)

The spColumn "Diagrams" module is a powerful tool especially for investigating interaction diagrams (failure surfaces) for columns and walls sections subjected to a combined axial force and biaxial bending moments. The module allows the user to view and analyze 2D interaction diagrams and contours along with 3D failure surfaces in a multi viewport environment. Figure 26 shows three views of:

1. P-M interaction diagram cut at angle of $23^{\circ}$
2. $\quad \mathrm{M}_{\mathrm{x}}-\mathrm{M}_{\mathrm{y}}$ interaction diagram cut at axial load of 1846.0 kip in compression
3. A 3D failure surface (interaction diagram) showing the points calculated in this example.

Figure 27 and Figure 28 show 3D visualization of failure surface with a horizontal and vertical plane cut, respectively.


Figure 26 - Diagrams Module (spColumn)

$\underline{\text { Figure } 27-3 D ~ V i s u a l i z a t i o n ~ o f ~ F a i l u r e ~ S u r f a c e ~ w i t h ~ a ~ H o r i z o n t a l ~ P l a n e ~ C u t ~ a t ~} \mathrm{P}=1846.0$ kip (spColumn)


Figure 28 - 3D Visualization of Failure Surface with a Vertical Plane Cut at $23^{\circ}$ (spColumn)


Figure 29 - 3D Visualization of Failure Surface with a Vertical Plane Cut at $0^{\circ}$ (spColumn)


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